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Direct Numerical Simulation of Drag Reduction with Uniform Blowing over a Rough Wall

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Background

Turbulence

- Huge drag
- Environmental problems
- High operation cost
- How to control?







Flow control classification

(M. Gad-el-Hak, J. Aircraft, 2001)



- Uniform blowing

Uniform blowing (UB)

(Sumitani & Kasagi, *AIAA J.*, 1995 Kametani & Fukagata, *J. Fluid Mech.*, 2011)

Drag contribution in a channel flow with UB(/US)

$$C_{f} = \frac{12}{\text{Re}_{b}} + 12 \int_{0}^{2} (1 - y)(-\overline{u'v'}) dy - 12V_{w} \int_{0}^{2} (1 - y)\overline{u} dy$$

$$Viscous \qquad \text{Turbulent} \qquad \text{Convective (=UB/US)} \\ Contribution \qquad \text{contribution} \qquad \text{contribution} \qquad \text{(Fukagata et al., Phys. Fluids, 2002)}$$

- Excellent performance (about 45% by $V_w = 0.5\% U_{\infty}$)
- Unknown over a rough wall



On a boundary layer, White: vortex core, Colors: wall shear stress

Investigate the interaction between roughness and UB for drag reduction

- DNS of turbulent channel flow
- Focus on drag reduction performance and mechanism

Numerical procedure

- Based on FD code (for wall deformation) (Nakanishi et al., Int. J. Heat Fluid Fl., 2012)
- Constant flow rate, $\operatorname{Re}_b = 2U_b \delta / \nu = 5600$
 - so that $Re_{\tau} \approx 180$ in a plane channel (K.M.M.)
- $\Delta x^+ = 4.4, 0.93 < \Delta y^+ < 6, \Delta z^+ = 5.9$
- UB magnitude: *M* = 0, 0. 001, 0. 005



Model of rough wall

(E. Napoli et al., J. Fluid Mech., 2008)

Roughness displacement

$$d(x) = \delta \sum_{i=1}^{4} A_i \sin\left(\frac{2^i \pi x}{L_x/2}\right)$$



Coordinate transformation

(S. Kang & H. Choi, Phys. Fluids, 2000)

Calculation grids: ξ_i (Cartesian with extra force)

$$\begin{cases} x = \xi_1 \\ y = \xi_2(1 + \eta) + \eta_d \\ z = \xi_3 \end{cases}$$

(x, y, z: physical coordinate)
$$\eta \equiv (\eta_u - \eta_d)/2 = -d(x)/2 \\ \eta_d = r(x), \eta_u = 0$$

$$\eta_d, \eta_u: \text{displacement of}$$

lower/upper wall

Actual grid points allocation



Post processing

Drag coefficient decomposition for rough case

$$C_{Duf} = \frac{8}{\text{Re}_{b}} \frac{d\bar{u}}{dy}\Big|_{\xi_{2}=2}$$

$$C_{Dlf} = \frac{8}{\text{Re}_{b}} \overline{\left(\frac{du}{dy}\Big|_{\xi_{2}=0} + \frac{dv}{dx}\Big|_{\xi_{2}=0}\right)}$$
(Friction component)
$$C_{Dlp} = -16 \frac{dP}{d\xi_{1}} - (C_{Dlf} + C_{Duf})$$
(Pressure component)

Drag reduction rate

$$R_{Dl} = \frac{\Delta C_D}{C_{D,M=0}} \times 100 \, [\%]$$

Only focusing on lower side, subscript "*l*" omitted hereafter

Drag reduction rate, R_D



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How does friction drag decrease?

Bulk mean streamwise velocity



How does pressure drag decrease?



Stream function



In practical applications



Concluding remarks

DNS of turbulent channel flow over a rough wall with UB

- UB is effective over rough walls
 - Lower drag reduction rate (7%, 26% / 11%, 37% in rough / smooth case, with M = 0.001, 0.005)
- Drag reduction mechanism
 - Friction drag by wall-normal convection (=conventional)
 - Pressure drag by prevention of stagnant flow
- Outlook toward practical applications
 - More saving opportunity over rough walls

Future plans

- Another drag reduction technique on rough surface
 - Spanwise oscillation (ongoing)
- Assessment of net energy? (should be external flow)
- Calculation at higher Reynolds number?
- Other types of rough surfaces (e.g., 3D structure)?

Uniform blowing (UB)

(Sumitani & Kasagi, *AIAA J.*, 1995 Kametani & Fukagata, *J. Fluid Mech.*, 2011)

Drag contribution in a channel flow with UB(/US)



- Excellent performance (about 45% by $V_w = 0.5\% U_{\infty}$)
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White: vortex core, Colors: wall shear stress

Governing equations

(S. Kang & H. Choi, Phys. Fluids, 2000)

Incompressible Continuity and Navier-Stokes in ξ_i coordinate

$$\begin{cases} \frac{\partial u_i}{\partial \xi_i} = -S \\ \frac{\partial u_i}{\partial t} = -\frac{\partial (u_i u_j)}{\partial \xi_j} - \frac{\partial p}{\partial \xi_i} + \frac{1}{\operatorname{Re}_b} \frac{\partial^2 u_i}{\partial \xi_j \xi_j} - \frac{dP}{d\xi_1} \delta_{i1} + S_i \end{cases}$$

where

Discretization methods

- Energy-conservative second-order finite difference schemes (In space)
- Low-storage third-order Runge-Kutta / Crank-Nicolson scheme (In time)
 - + SMAC method for pressure correction

Discretized in the staggered grid system

Validation & Verification

(B. Milici et al., J. Fluid Mech., 2014)



Bulk mean streamwise velocity

Time trace of instantaneous C_{D,r}

Less than 2% of difference with most resolved one

How does friction drag decreases?



*averaged in the spanwise and time

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\overline{v} distribution (2D contour)

Based on u_{τ} in w/o control case

UB 0.5% case



No control case

\overline{u} distribution (2D contour)

Based on u_{τ} in w/o control case

UB 0.5% case



No control case

Stream function (detailed)



$u_{\rm rms}$ distribution



Black: w/o control Green: UB 0.1% Red: UB 0.5%

$v_{\rm rms}$ distribution



Black: w/o control Green: UB 0.1% Red: UB 0.5%

$w_{\rm rms}$ distribution



Black: w/o control Green: UB 0.1% Red: UB 0.5%

$\overline{u'^+v'^+}$ distribution



Black: w/o control Green: UB 0.1% Red: UB 0.5%

- Effective slope = about 0.2
- k_rms
- Sand grain roughness
- 3D?
- How to calculate skin-friction drag
- The history of UB
- Pressure component legend (purple) of smooth case should be removed
- More time for drag reduction rate slide