

THE EFFECTS OF POROUS WALLS ON TRANSITIONAL AND TURBULENT CHANNEL FLOWS

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Background	
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Setup

Stability

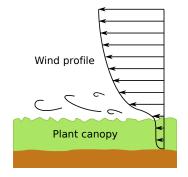
Turbulence

Conclusions

WHY POROUS WALLS?

Flow over porous media are common:

- Water-immersed surfaces
- Biologic surfaces (blood vessels, vascular protheses)
- Atmospheric boundary layer over plant canopies
- Aeronautics



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STATE OF THE ART

STABILITY: POROUS WALLS ARE DE-STABILIZING

- Critical linear Reynolds number decreases by 90%
- Sparrow (JAM 1973), Tilton & Cortelezzi (JFM 2008)

TURBULENT FLOWS

- Porous layer is neglected and represented via a wall boundary condition
- Very simple porous material are considered
- Present approach: solve VANS equations

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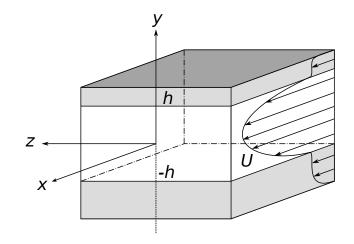
Stability

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PROBLEM SETUP

TWO BOUNDARY LAYERS ACROSS EVERY POROUS LAYER



Background	Setup	Stability	Turbulence	Conclusions

MODEL OF THE POROUS MEDIUM

Porosity ε

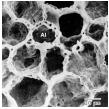
- · Volume fraction of the fluid phase
- $\varepsilon = V_{fluid}/V_{total}, 0 \le \varepsilon \le 1$

PERMEABILITY K [M²]

- The ease of fluid flow through the medium
- Non-dimensional permeability $\sigma = \sqrt{K}/h, \ \sigma \ge 0$



Alveoli in the lungs



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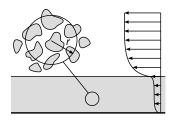
THE VOLUME-AVERAGED NAVIER–STOKES EQUATIONS VANS: WHITAKER, 1996

SIMPLIFIED VANS: LINEAR!

$$\nabla \cdot \langle \mathbf{u} \rangle = 0$$
$$\frac{\partial \langle \mathbf{u} \rangle}{\partial t} = -\frac{1}{\rho} \nabla \langle p \rangle + v \nabla^2 \langle \mathbf{u} \rangle - \frac{v}{K} \varepsilon \langle \mathbf{u} \rangle$$

 $\langle {f u}
angle$ and $\langle p
angle$ are continuous functions

- Homogeneous porosity
- Isotropic porosity
- Inertial effects negligible



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Background Setup Stability Turbulence

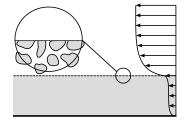
OCHOA-TAPIA AND WHITAKER, IJHMT 1995

CONTINUITY

$$\mathbf{u}=\langle \mathbf{u}\rangle,\,p=\langle p\rangle$$

STRESS JUMP CONDITIONS

$$\frac{1}{\varepsilon} \frac{\partial \langle u \rangle}{\partial y} - \frac{\partial u}{\partial y} = \pm \frac{\tau}{\sqrt{K}} u$$
$$\frac{1}{\varepsilon} \frac{\partial \langle w \rangle}{\partial y} - \frac{\partial w}{\partial y} = \pm \frac{\tau}{\sqrt{K}} w$$



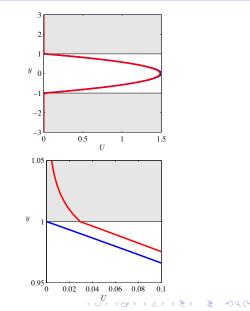
- The stress jump models the momentum transfer
- *τ* accounts for the surface manufacturing/machining

Conclusions

Background	Setup	Stability	Turbulence	Conclusions
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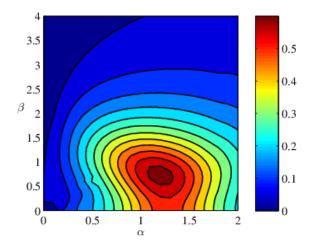
NON-MODAL STABILITY

- Analytical base flow
- Chebyshev discretization
- 7 parameters: *Re*, α, β, h_p, σ, ε, τ
- Adaptive algorithm, >4M cases



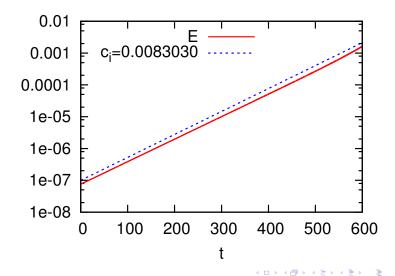
Background	Setup	Stability	Turbulence	Conclusions
	LARGEST FRA	ACTIONAL C	HANGE IN G_{max}	

 $Re = 500, h_p = 0.5, \sigma = 0.02, \varepsilon = 0.3, \tau = 0.5$





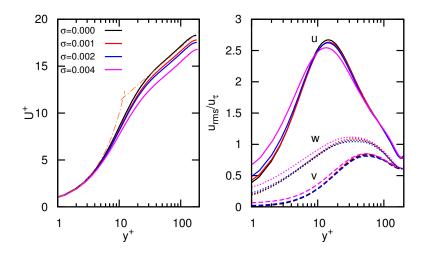
 $Re = 2800, \varepsilon = 0.6, \sigma = 0.004, \tau = 0, h_p = 1$



Background	Setup	Stability	Turbulence	Conclusions

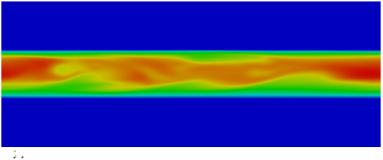
LIMIT OF SOLID WALL

 $Re = 2800, \varepsilon = 0.6, \tau = 0, h_p = 0.1$



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CONCLUSIVE REMARKS

- Non-modal stability
- Turbulent flow
- Ongoing work: control, drag reduction