

# NEAR-WALL ENSTROPY GENERATION IN A DRAG-REDUCED TURBULENT CHANNEL FLOW WITH SPANWISE WALL OSCILLATIONS

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## ACTIVE OPEN-LOOP TECHNIQUE

Energy input into system  
Pre-determined forcing

## NUMERICAL APPROACH

Direct numerical simulations of wall turbulence  
Fully-developed turbulent channel flow ( $Re_\tau = u_\tau h / \nu = 200$ )  
Compact finite-difference scheme along wall-normal direction  
Spectral discretization along streamwise and spanwise directions

## SPANWISE WALL OSCILLATIONS

- New approach: *Turbulent enstrophy*
- *Transient evolution*

## CONSTANT DP/DX

$\tau_w$  is fixed in fully-developed conditions

**GAIN:**  $U_b$  increases

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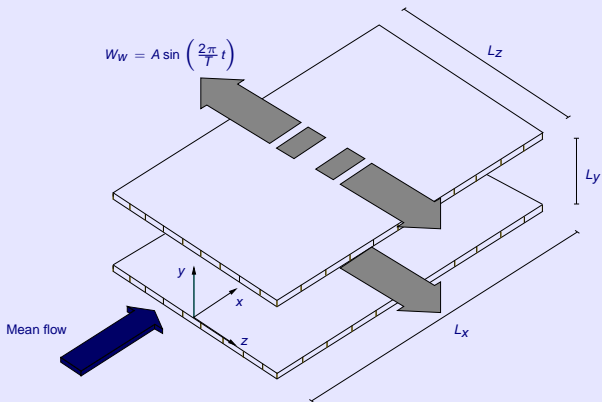
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# SPANWISE WALL OSCILLATIONS

## GEOMETRY

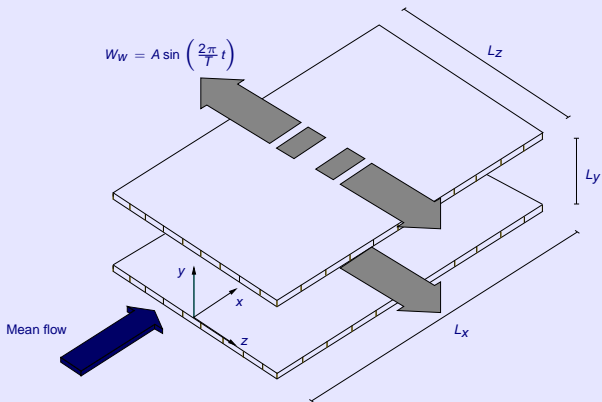


$$R = \frac{C_{f,r} - C_{f,o}}{C_{f,r}} = \frac{U_{b,o}^2 - U_{b,r}^2}{U_{b,o}^2}$$

Why does the skin-friction coefficient decrease?

$C_f = 2\tau_w / (\rho U_b^2)$  decreases  $\rightarrow$  study why  $U_b$  increases

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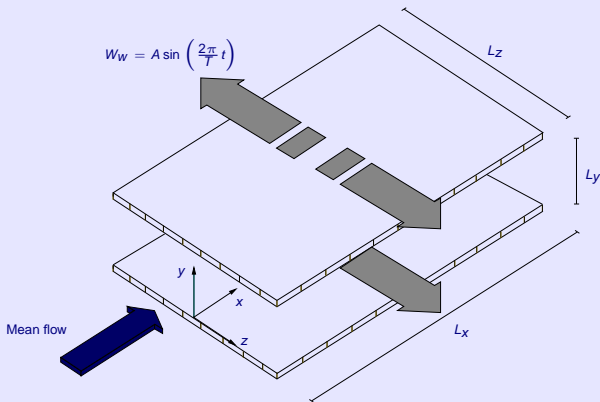
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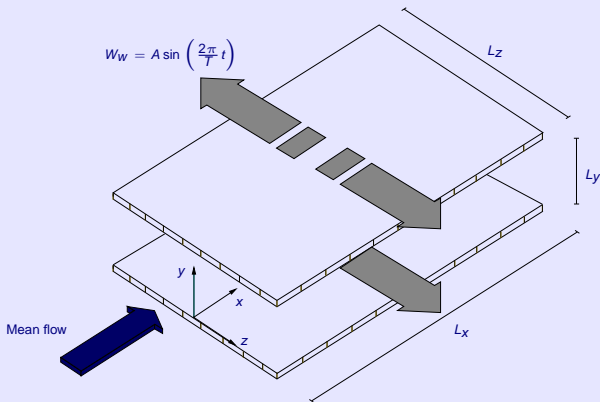
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SPACE: HOMOGENEOUS DIRECTIONS

$$\bar{f}(y, t) = \frac{1}{L_x L_z} \int_0^{L_x} \int_0^{L_z} f(x, y, z, t) dz dx$$

PHASE

$$\hat{f}(y, \tau) = \frac{1}{N} \sum_{n=0}^{N-1} \bar{f}(y, nT + \tau)$$

TIME

$$\langle f \rangle (y) = \frac{1}{T} \int_0^T f(y, \tau) d\tau$$

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$$[f]_g = \int_0^h \langle f \rangle (y) dy$$

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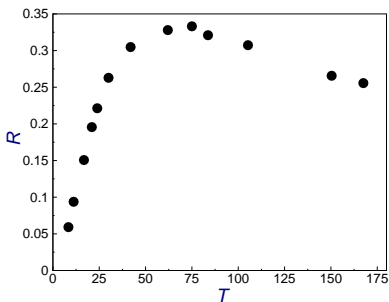
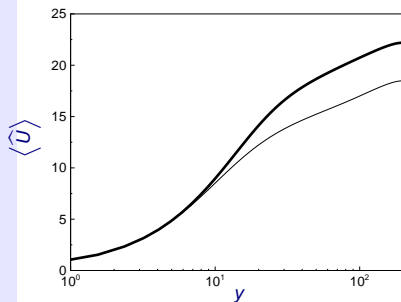
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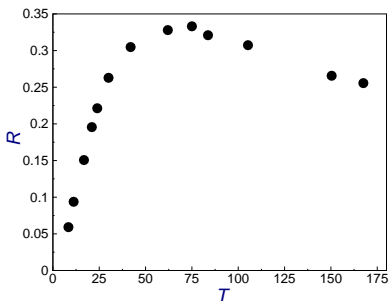
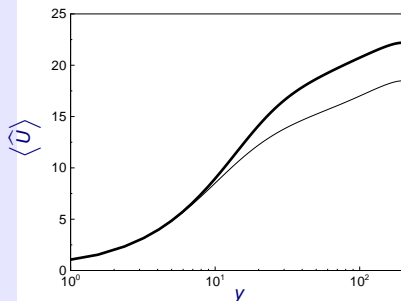
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Mean velocity increases in the bulk of the channel

Mean wall-shear stress is unchanged

Optimum period of oscillation  $T \approx 75$



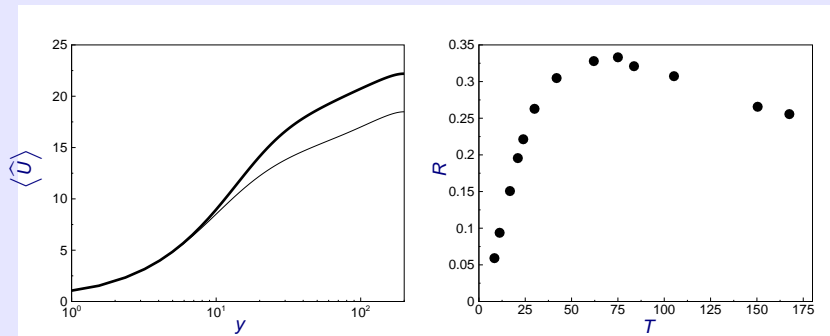


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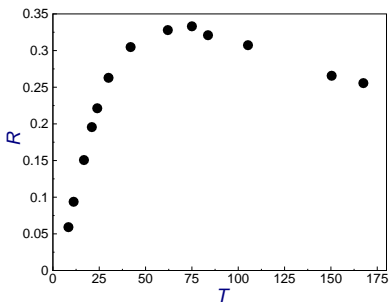
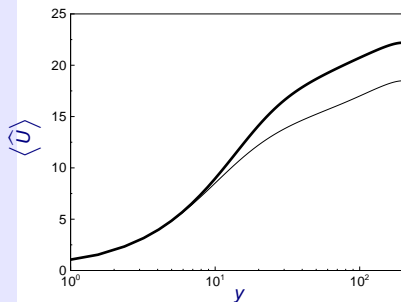


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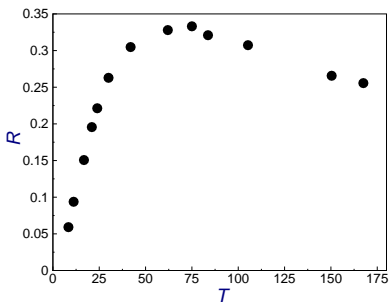
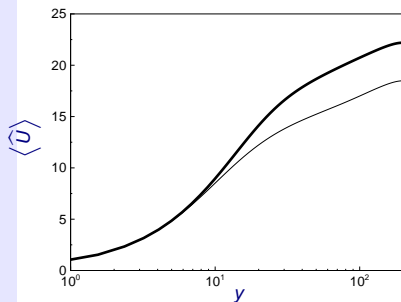


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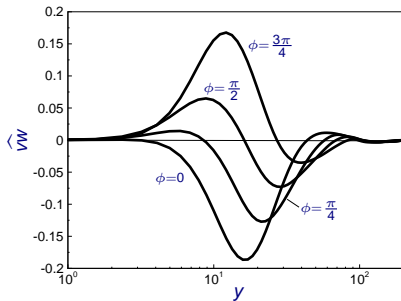
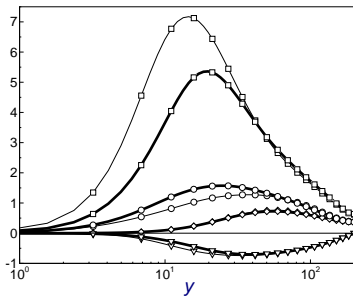
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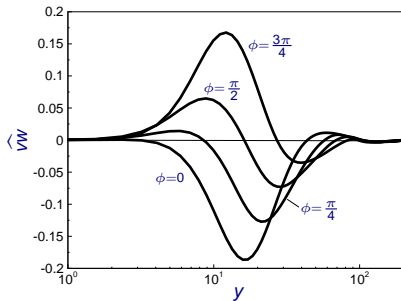
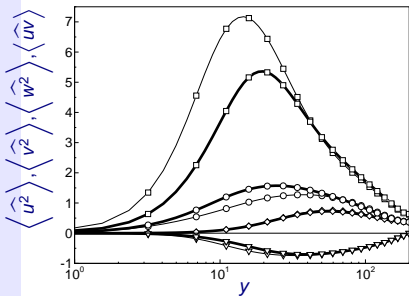
$\langle \widehat{u}^2 \rangle$ ,  $\langle \widehat{v}^2 \rangle$ ,  $\langle \widehat{w}^2 \rangle$ ,  $\langle \widehat{uv} \rangle$



Turbulence kinetic energy decreases

Streamwise velocity fluctuations are attenuated the most

New oscillatory Reynolds stress term  $\widehat{vw}$  is created,  $\langle \widehat{vw} \rangle = 0$

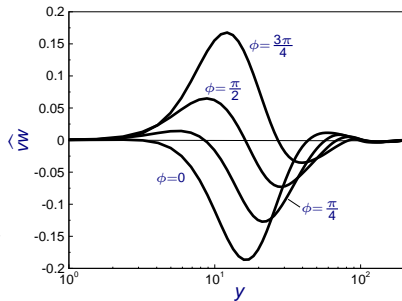
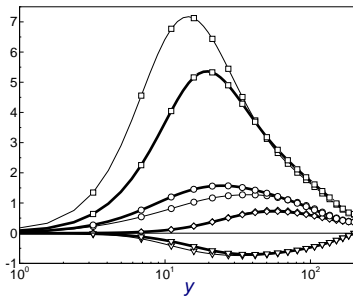


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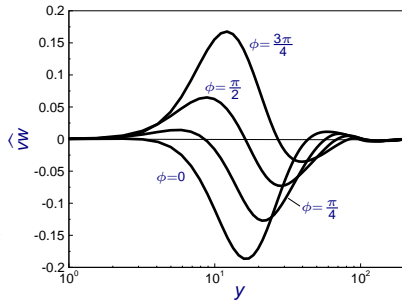
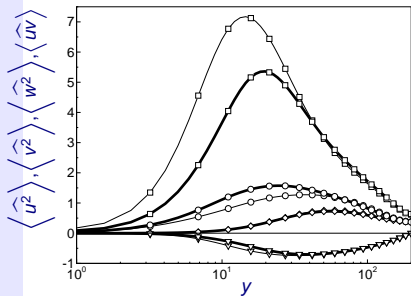
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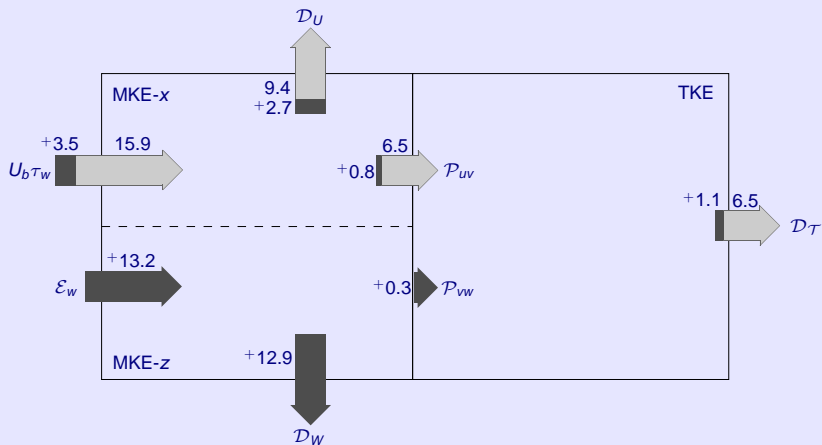
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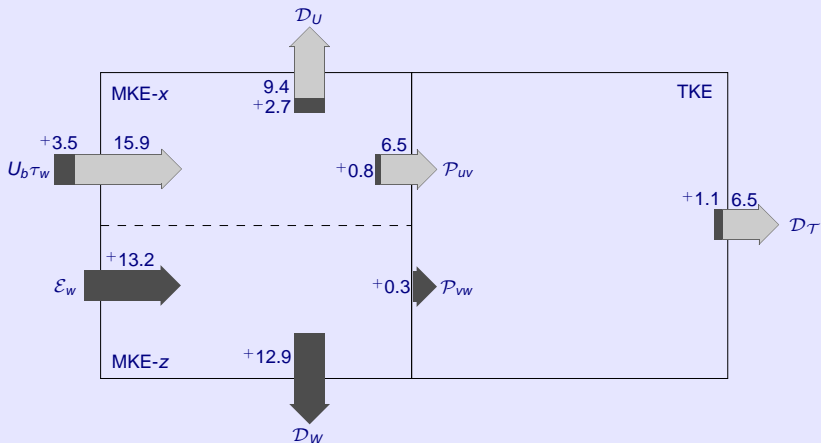
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Energy is dissipated through:

Mean-flow viscous effects ( $\rightarrow \mathcal{D}_U, \mathcal{D}_W$ )

Turbulent viscous effects ( $\rightarrow \mathcal{D}_T$ )

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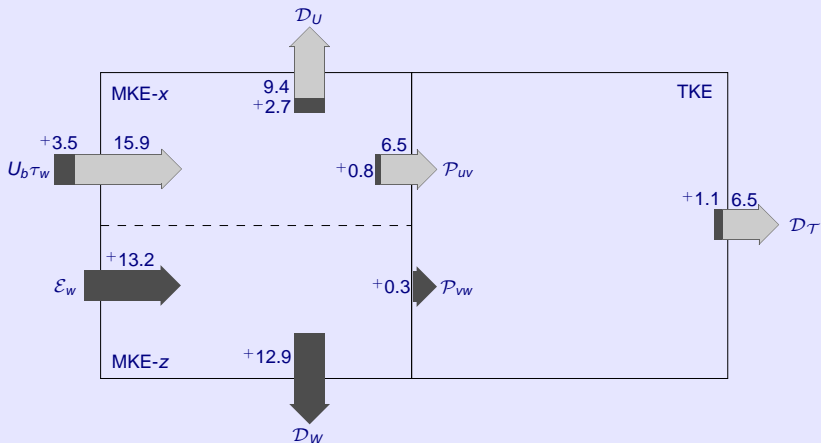


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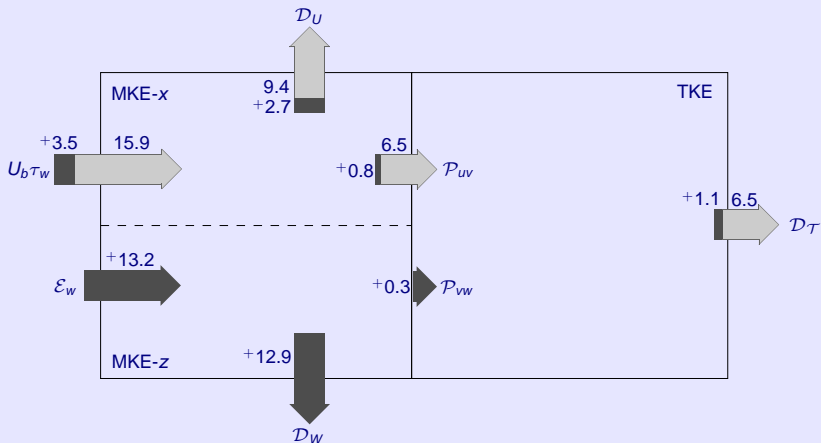
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## GLOBAL MEAN KINETIC ENERGY EQUATION

$$U_b \tau_w + \underbrace{\left\langle A \frac{\partial \widehat{W}}{\partial y} \Big|_{y=0} \right\rangle}_{\varepsilon_w} = - \underbrace{\left[ \widehat{uv} \frac{\partial \widehat{U}}{\partial y} \right]_g}_{\mathcal{P}_{uv}} - \underbrace{\left[ \widehat{vw} \frac{\partial \widehat{W}}{\partial y} \right]_g}_{\mathcal{P}_{vw}} + \underbrace{\left[ \left( \frac{\partial \widehat{U}}{\partial y} \right)^2 \right]_g}_{\mathcal{D}_U} + \underbrace{\left[ \left( \frac{\partial \widehat{W}}{\partial y} \right)^2 \right]_g}_{\mathcal{D}_W}$$

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### TOTAL KINETIC ENERGY BALANCE

$$U_b \tau_w + \varepsilon_w = \mathcal{D}_U + \mathcal{D}_W + \mathcal{D}_T$$

### TURBULENT DISSIPATION

$$\mathcal{D}_T = \left[ \widehat{\omega_j \omega_j} \right]_g$$

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## STILL TO BE ANSWERED

Why does TKE decrease?

Why does  $U_b$  increase?

## THREE POSSIBILITIES

- 1 Stokes layer acts on  $\mathcal{D}_U$  directly

→ excluded because  $W$  does not work directly on  $(\partial \hat{u} / \partial y)^2$

- 2 Stokes layer acts on  $\mathcal{P}_{uv}$  directly

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- 3 Stokes layer acts on  $\mathcal{D}_T = [\widehat{\omega_j \omega_l}]_g$  directly

→  $W$  works on turbulent vorticity transport

## TURBULENT ENSTROPY TRANSPORT

Study the transport of turbulent enstrophy  $\widehat{\omega_j \omega_l}$

The term *enstrophy* was coined by G. Nickel and is from Greek  $\sigma\tau\rho\phi\acute{\eta}$ , which means *turn*

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Study the transport of turbulent enstrophy  $\widehat{\omega_j \omega_l}$

The term *enstrophy* was coined by G. Nickel and is from Greek  $\sigma\tau\rho\phi\acute{\eta}$ , which means *turn*

## STILL TO BE ANSWERED

Why does TKE decrease?

Why does  $U_b$  increase?

## THREE POSSIBILITIES

- 1 Stokes layer acts on  $\mathcal{D}_U$  directly

→ excluded because  $W$  does not work directly on  $(\partial \hat{U} / \partial y)^2$

- 2 Stokes layer acts on  $\mathcal{P}_{uv}$  directly

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- 3 Stokes layer acts on  $\mathcal{D}_T = [\widehat{\omega_j \omega_j}]_g$  directly

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$$\begin{aligned}
 \underbrace{\frac{1}{2} \frac{\partial \widehat{\omega_i \omega_j}}{\partial \tau}}_1 &= \underbrace{\widehat{\omega_x \omega_y} \frac{\partial \widehat{U}}{\partial y}}_2 + \underbrace{\widehat{\omega_z \omega_y} \frac{\partial \widehat{W}}{\partial y}}_3 + \underbrace{\widehat{\omega_j} \frac{\partial u}{\partial x_j} \frac{\partial \widehat{W}}{\partial y}}_4 - \underbrace{\widehat{\omega_j} \frac{\partial w}{\partial x_j} \frac{\partial \widehat{U}}{\partial y}}_5 \\
 &- \underbrace{\widehat{v \omega_x} \frac{\partial^2 \widehat{W}}{\partial y^2}}_6 + \underbrace{\widehat{v \omega_z} \frac{\partial^2 \widehat{U}}{\partial y^2}}_7 + \underbrace{\widehat{\omega_j \omega_j} \frac{\partial u_i}{\partial x_j}}_8 - \underbrace{\frac{1}{2} \frac{\partial}{\partial y} (\widehat{v \omega_j \omega_j})}_9 \\
 &+ \underbrace{\frac{1}{2} \frac{\partial^2 \widehat{\omega_j \omega_j}}{\partial y^2}}_{10} - \underbrace{\frac{\partial \widehat{\omega_i}}{\partial x_j} \frac{\partial \widehat{\omega_j}}{\partial x_j}}_{11}
 \end{aligned}$$

Stokes layer influences dynamics of turbulent enstrophy

Three terms: which is the dominating one?

→ Let's look at the terms of the equation



$$\begin{aligned}
 \underbrace{\frac{1}{2} \frac{\partial \widehat{\omega_i \omega_j}}{\partial \tau}}_1 &= \underbrace{\widehat{\omega_x \omega_y}}_2 \frac{\partial \widehat{U}}{\partial y} + \underbrace{\widehat{\omega_z \omega_y}}_3 \frac{\partial \widehat{W}}{\partial y} + \underbrace{\widehat{\omega_j \frac{\partial u}{\partial x_j}}}_4 \frac{\partial \widehat{W}}{\partial y} - \underbrace{\widehat{\omega_j \frac{\partial w}{\partial x_j}}}_5 \frac{\partial \widehat{U}}{\partial y} \\
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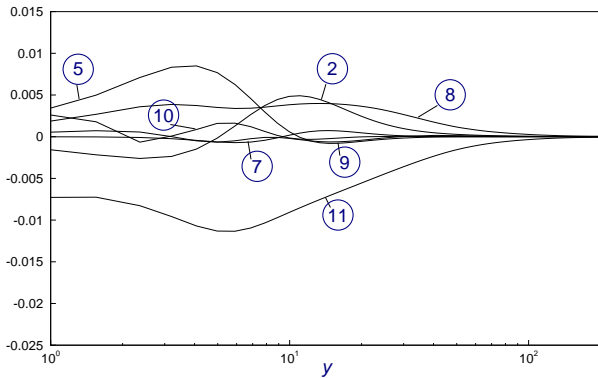
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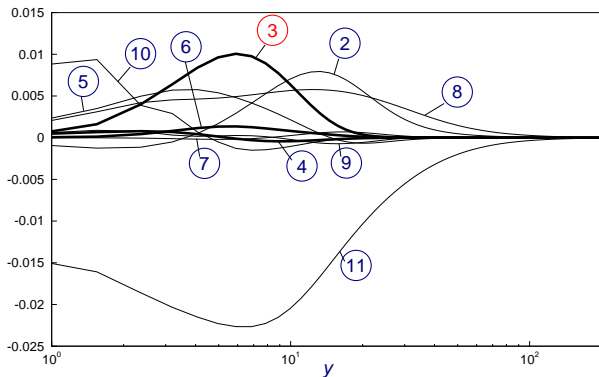
# TURBULENT ENSTROPY PROFILES

FIXED WALL



# TURBULENT ENSTROPY PROFILES

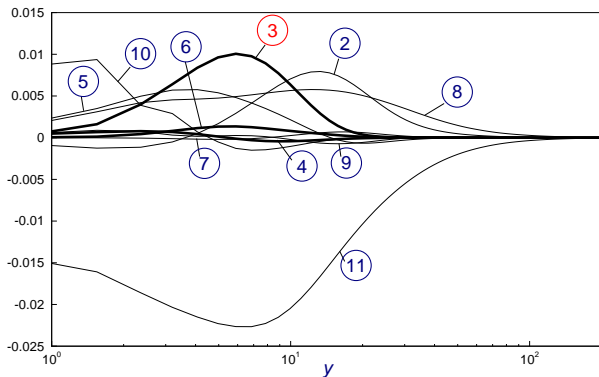
## OSCILLATING-WALL PROFILES



Term 3,  $\widehat{\omega_z \omega_y} \partial \widehat{W} / \partial y \rightarrow$  turbulent enstrophy production is dominant  
Other oscillating-wall terms are much smaller  
Turbulent dissipation of turbulent enstrophy increases

# TURBULENT ENSTROPY PROFILES

## OSCILLATING-WALL PROFILES



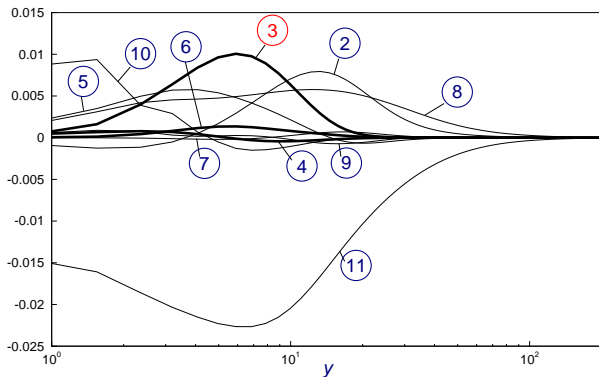
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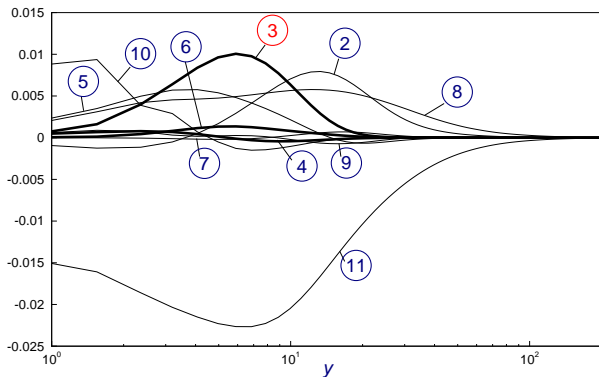
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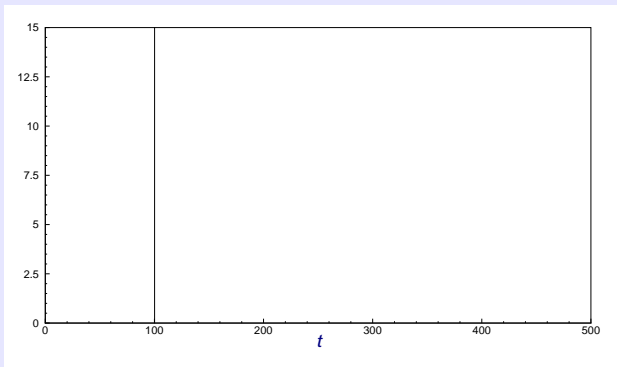
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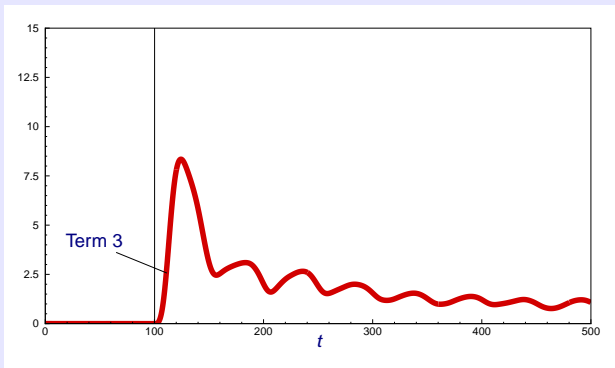
Key: transient from start-up of wall motion



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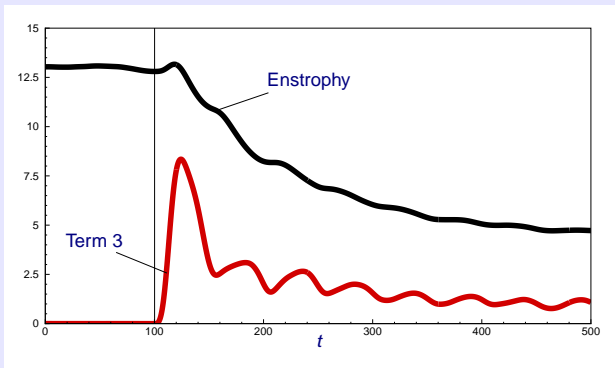


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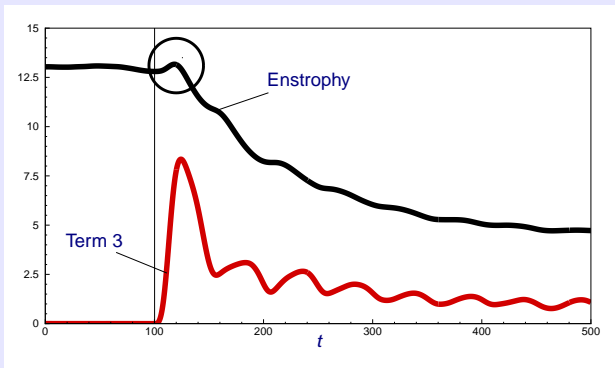
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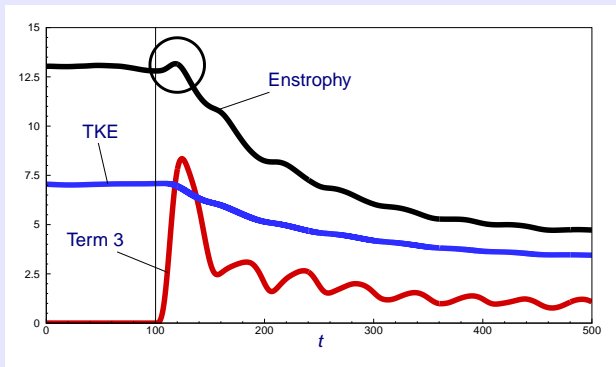
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## USEFUL INFORMATION

**RED:** term 3 increases abruptly, then decreases

**BLACK:** turbulent enstrophy increases, then decreases

**BLUE:** TKE decreases monotonically

## SHORT STAGE

Turbulent enstrophy increases through  $\widehat{\omega_z \omega_y} \partial \widehat{W} / \partial y$

## INTERMEDIATE STAGE

TKE decreases because of enhanced turbulent dissipation

## LONG STAGE

Bulk velocity increases because of TKE reduction

→ drag reduction

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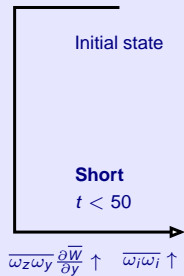
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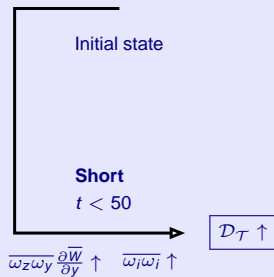
## LONG STAGE

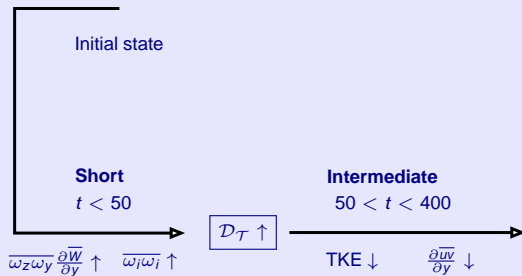
Bulk velocity increases because of TKE reduction

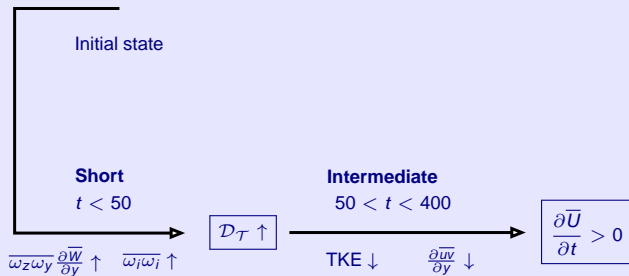
→ drag reduction

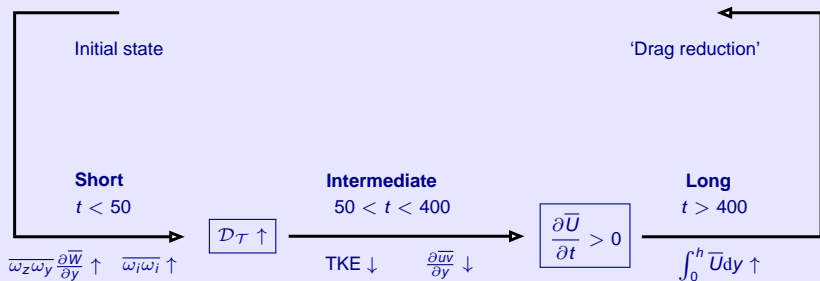
Initial state





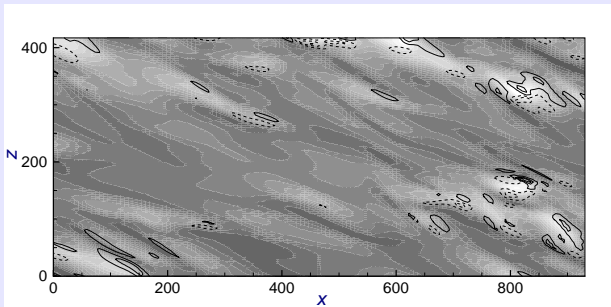




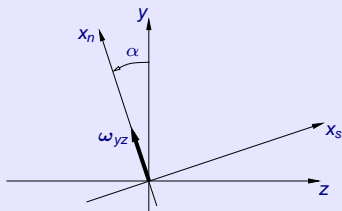




- $\widehat{\omega_z \omega_y} \partial \widehat{W} / \partial y$  is key term leading to drag reduction
- $\widehat{\omega_z \omega_y} \partial \widehat{W} / \partial y \rightarrow \partial \widehat{W} / \partial y$  acts on  $\widehat{\omega_z \omega_y}$
- $\widehat{\omega_z \omega_y} \approx \frac{\partial u}{\partial y} \frac{\partial u}{\partial z}$ 
  - $\frac{\partial u}{\partial y} \rightarrow$  upward eruption of near-wall low-speed fluid
  - $\frac{\partial u}{\partial z} \rightarrow$  lateral flanks of the low-speed streaks



$\frac{\partial u}{\partial y} \frac{\partial u}{\partial z}$  located at the sides of high-speed streaks



## SIMPLIFIED TURBULENT ENSTROPY EQUATION

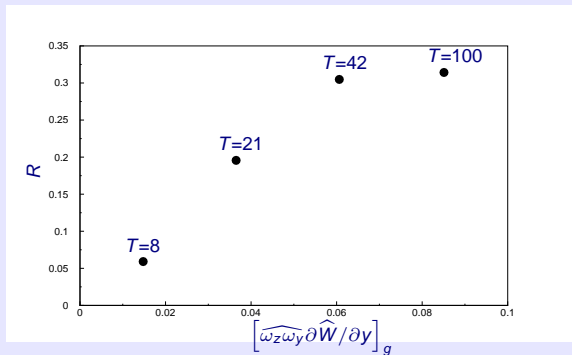
$$\frac{1}{2} \frac{\partial}{\partial t} (\omega_y^2 + \omega_z^2) = \omega_z \omega_y G - \left( \frac{\partial \omega_y}{\partial y} \right)^2 - \left( \frac{\partial \omega_z}{\partial y} \right)^2$$

Rotation of axis

$$\frac{1}{2} \frac{\partial \omega_n^2}{\partial t} = S_{nn} \omega_n^2 - \left( \frac{\partial \omega_n}{\partial y} \right)^2$$

Integration by Charpit's method

$$\omega_n = \omega_{n,0} \underbrace{e^{\sin \alpha \cos \alpha G t}}_{\text{stretching}} \underbrace{e^{-\beta^2 t} e^{-\beta y}}_{\text{dissipation}}, \quad \beta = \frac{\partial \omega_n / \partial t}{\partial \omega_n / \partial y} \sim \frac{\lambda_y}{\lambda_t}$$



Drag reduction grows monotonically with global production term  
 This happens up to optimum period

THANK YOU!

#### REFERENCE

Ricco, P. Ottonelli, C. Hasegawa, Y. Quadrio, M.  
Changes in turbulent dissipation in a channel flow with oscillating walls  
*J. Fluid Mech.*, 700, 77-104, 2012.